P4EA Revision 13 Sensenich 76E Series August 2, 2019

## TYPE CERTIFICATE DATA SHEET NO. P4EA

Propellers of models described herein conforming with this data sheet (which is part of Type Certificate No. P4EA) and other approved data on file with the Federal Aviation Administration, meet the minimum standards for use in certificated aircraft in accordance with pertinent aircraft data sheets and applicable portions of the Federal Aviation Regulations provided they are installed, operated and maintained as prescribed by the approved manufacturer's manuals and other approved instructions.

Type Certificate Holder Sensenich Propeller Manufacturing Company, Inc.

14 Citation Lane Lititz, PA 17543

Type Fixed-Pitch Metal Material Aluminum Alloy

Number of Blades Two

Basic Model Takeoff &			Standard		Hub Drilling		Hub Dimensions		Weight	
(See Note 2)	Max. HP	. Cont. RPM	Diameter	Pitch	No. Holes	Dia. Holes	Dia. Bolt Circle	Dia.	Thick.	(lb.) (Max. Dia.)
76EM8	180	2700	76"	68"-52"	6	33/64"	4-3/4"	6"	3-9/16"	34.5
76EC	210	2800	76"	68"-52"	6	33/64"	4"	6"	3-9/16"	34.5
Certification E	Basis			Type Cert	ificate N	lo. P4EA	cember 15, 1 issued May e Certificate	14, 1962	er 7, 1961	

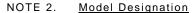
For Note 9 Superior Vantage O-360 / IO-360 Series Engines, FAR 35, Amendment 35-7, Dated Decmber 28, 1995

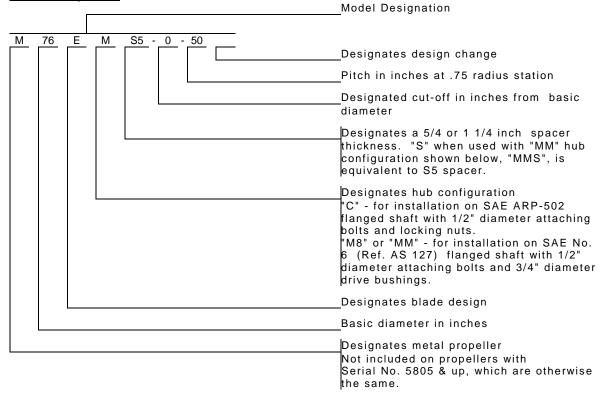
For Note 9 Continental IO-370 D Series Engines, FAR 35, Amendment 35-7, Dated December 28, 1995

Production Basis Production Certificate No. 1NE

- NOTE 1. Installation. These models are for installation on flanged propeller shaft ends (see NOTE 2). Installation is to be made with special steel bolts which are furnished or specified by the propeller manufacturer in accordance with the appropriate propeller assembly drawing. See NOTE 7 for spacer designations and NOTE 9 for approved spacer lengths.
  - a. Propeller Model 76EM8 is installed on SAE No. 2 flanged shaft with 1/2 inch diameter attaching bolt and 3/4 inch diameter drive bushings.
  - b. Propeller Model 76EC is installed on SAE ARP-502 flanged shaft with 1/2 inch diameter attaching bolt and locking nuts.

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NOTE 3, 4, 5, 6, and 8. Not applicable.

## NOTE 7. Accessories:

<u>Spacers</u>. Sensenich spacer models are identified by flange codes (see NOTE 2) and spacer thickness designated based on multiples of 1/4 inch. See NOTE 9 for approved spacer lengths.

Spinners. The following spinner parts are approved for use with propeller model 76EC:

Sensenich Dwg No.	<u>Description</u>
C-2347	Spinner Dome
C-2391	Rear Bulkhead
C-2392	Front Bulkhead

NOTE 9. <u>Special Limits.</u> <u>Table of Propeller-Engine Combinations</u>
<u>Approved Vibrationwise for Use on Normal Category Single-Engine Tractor Aircraft</u>

The maximum and minimum propeller diameters that can be used from a vibration standpoint are shown below. No reduction below the minimum diameter listed is permissible since this figure includes the diameter reduction allowable for repair purposes.

Propeller Model	Engine Model	Max. Dia. (Inches)	Min. Dia. (Inches)	Placards
76EM8, Spacers 0 to 4.0 inches, incl.	Lycoming O-360/IO-360 Series with Hollow Crankshafts (Excludes engines with suffixes having a digit "4" or higher in the second position.) 180 h.p. and 2700 r.p.m.	76	76	Avoid continuous operation between 2150 and 2350 r.p.m.
76EM8, Spacers 0 to 4.0 inches, incl.	Lycoming O-360/IO-360 Series with Solid Crankshafts (Engines with suffixes having a digit "4" or higher in the second position.) 180 h.p. and 2700 r.p.m.	76	76	None
76EM8, Spacers 0 to 4.0 inches, Incl	Superior VantageO-360 / IO-360 Series Engines with Hollow Crankshafts 180h.p. and 2700 r.p.m.	76	76	Avoid continuous operation between 2100 and 2350 r.p.m
76EM8 Spacers 0 to 3.5 inches, incl.	Continental IO-370 D Series Engines with Solid Crankshafts 185 h.p. and 2700 r.p.m.	76	76	None.
76EC, Spacers 0 to 4.0 inches, incl.	Continental IO-360-ES Series engines 210 h.p. and 2800 r.p.m.	76	74	Avoid continuous static operation above 2100 r.p.m.

NOTE 10. <u>Special Notes.</u> The word "eligible" as used herein does not signify approval. For approval, compliance with the applicable aircraft airworthiness requirements is necessary.

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